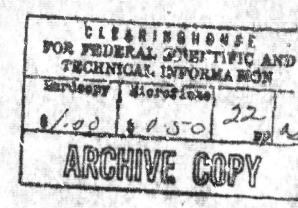
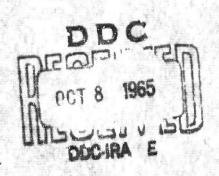
MEMORANDUM RM-3773-ARPA AUGUST 1985



ALTERNATIVE TECHNIQUES FOR THE INVARIANT IMBEDDING OF RAREFIED COUETTE FLOWS

J. Aroesty, R. Beliman, R. Kalaba and S. Ueno

PREPARED FOR ADVANCED RESEARCH PROJECTS AGENCY





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PREFACE

Invariant imbedding techniques have proven useful in the computational solution of problems in neutron transport and radiative transfer. This Memorandum discusses an application of these techniques to a problem in rarefied gas dynamics and should be of interest to aerodynamicists, physicists, and applied mathematicians.

SUMMARY

This Memorandum re-examines the application of invariance techniques to plane Couette-flow problems and presents several different approaches to the derivation of the invariant-imbedding equations.

It is shown how both linear and nonlinear collision operators may be considered, for both discrete and continuous versions of the original Boltzmann formulation.

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SYMBOLS

c velocity vector, defined by three velocity components u_{i_u} , v_{i_v} , w_{i_w}

 $\mathbf{f_i}$ = distribution function for particles with velocity vector $\mathbf{C_i}$

K_i = collision operator for particles in ith velocity
state

L = linearized collision operator

L+ = linearized collision operator acting on 4+

L = linearized collision operator acting on \$\delta\$

R = reflection function

* R = nonlinear reflection function

 r_i^+ , w_i^- = defined by Eq. (20)

S = one-dimensional reflection function

u, v, w = molecular velocities for particles in 4

u, c, w = velocity components in discrete velocity space

u, v, w = molecular velocities for particles in \$\\ ^{\bar{\circ}}

x = distance along normal to a moving plate

z = spacing between plates

γ = source function for molecules in i th state entering

perturbation distribution function

perturbation distribution function for particles with velocity components in the direction of increasing:

perturbation distribution function for particles with velocity components in the direction of decreasing x

I. INTRODUCTION

In an earlier paper, (1) the concepts of invariant imbedding were applied to a simple problem in rarefied gas dynamics, that of steady, linearized shear flow between two infinite plane walls under the assumptions of Krook scattering and diffuse reflection from both walls. Particle-counting techniques were applied directly to the Krook collision operator, reflection functions were introduced, and the characteristically nonlinear integro-differential equation of the initial-value type for the reflection function was derived. Subsequent work (2) has concentrated on the computational solution of that integral equation as well as on the unsteady version of this same linearized shear flow and on the effect of specular-diffuse molecular reflection from boundaries.

In this Memorandum, we briefly re-examine the application of invariance techniques to plane Couette-flow problems and present several alternate prescriptions for the derivation of the invariant-imbedding equations directly from the Boltzmann formulation. Our objective here is not to present any new results but rather to review and extend several techniques which have already been developed.

It is clear by now that a more formal and mechanical approach than particle counting is desirable for the derivation of invariance equations. Particle counting, unless performed with great care, can lead to errors in problems involving complex geometries and nonlinear scattering models. A rigorous treatment of the equations of invariant imbedding has recently been given by Bailey, (3) who considered a number of related linear problems in transport theory. In particular, he has re-examined the earlier analyses for spherical geometries and concludes that perturbation techniques are the most direct method of approach.

The invariant-imbedding approach involves the calculation of the effect of a variation of the spacing between the two boundaries for a fixed incoming distribution function of a particle at one boundary. We shall consider two different approaches to the problem; the first, that of Wing, (4) is suitable for linear problems, while the second, that of Bellman and Kalaba, (5) is suitable for both linear and nonlinear problems.

II. ANALYSIS

METHOD I

We consider a linearized Couette flow between two infinite planar walls. The Boltzmann equation for this flow becomes

$$u \frac{\partial \Phi}{\partial x} (u, v, w) = L(\Phi)$$
 (1)

The notation of Ref. 1 is employed, where u, v, and w are dimension-less molecular velocities, † is a perturbation distribution function, x is distance in units of mean free path, and L(†) is a linearized collision operator. It was shown in Ref. 2 how it is possible to uncouple the effect of the boundary-surface interaction by dealing with a standard so of boundary conditions corresponding to one wall being a sink for molecules and the other wall being a molecular source. This uncoupling is only valid for linearized flows. The solution of this "standard" problem can then be used to construct solutions which include the effects of nondiffuse surface-particle interaction. The standard boundary conditions are

$$\Phi^{+}(u, v, w, f(u_{0}, v_{0}, w_{0}), 0) = 0$$
 (sink at $x = 0$)
$$\Phi^{-}(u_{0}, v_{0}, w_{0}, f(u_{0}, v_{0}, w_{0}), z) = f(u_{0}, v_{0}, w_{0})$$
 (arbitrary source of molecules at $x = z$)

where we have chosen to indicate explicitly the dependence of the Φ 's on f(u, v, w). We seek a function $R(u, v, w, u_c, v_o, w_o)$, called a reflection function, such that

The subscripted variables u, v, w refer to the velocity space of particles with velocity components in the direction of decreasing x, or those particles in \$\phi\$, while the unsubscripted variables v, w refer to the velocity space of particles with velocity components in the direction of increasing x, or those particles in \$\psi^+\$.

$$\int_{-\infty}^{\infty} (u, v, w, f(u, v, w), z) = \int_{-\infty}^{0} \int_{-\infty}^{\infty} R(u, v, w, u_0, v_0, w_0, z)$$

$$f(u_o, v_o, w_o) du_o dv_o dw_o$$
 (3)

We consider the same problem as that given by Eq. (3) and boundary conditions of Eq. (2) on the interval $0 \le x \le z + \Delta$. Then

$$\Phi^{+}(u, v, w, f(u_{o}, v_{o}, w_{o}, 0) = 0$$

$$\Phi^{-}(u_{o}, v_{o}, w_{o}, f(u_{o}, v_{o}, w_{o}), z + \Delta) = f(u_{o}, v_{o}, w_{o})$$
(4)

and

$$\Phi^{+}(u, v, w, f(u_{o}, v_{o}, w_{o}), z + \Delta) = \int_{-\infty}^{0} \int_{-\infty}^{+\infty} R(u, v, w, u_{o}, v_{o}, w_{o}, z + \Delta)$$

$$f(u_0, v_0, w_0) du_0 dv_0 dw_0$$
 (5)

From Eq. (1), we can write

$$\phi^+(z + \Delta, f) =$$

$$\Phi^+(z, f) + \frac{1}{u} \Delta L^+(\Phi^+(u, v, w, f, z), \Phi^-(u_0, v_0, w_0, f, z) + o(\Delta))$$
 (6)

where

$$\lim_{\Delta \to 0} \frac{o(\Delta)}{\Delta} = 0$$

and $f = f(u_0, v_0, w_0)$. We now observe that if we consider this problem on the interval $0 \le x \le z$ we may write

$$f^{+}(u, v, w, f, z) = \int_{-\infty}^{0} \int_{-\infty}^{+\infty} \int_{-\infty}^{+\infty} R(u, v, w, u_{o}, v_{o}, w_{o}, z)$$

$$f^{+}(u_{o}, v_{o}, w_{o}) du_{o} dv_{o} dw_{o}$$
(7)

where $f^*(u_c, v_o, w_o)$ is the source function for the new problem.

$$f^{*}(u_{o}, v_{o}, w_{o}) = \Phi^{*}(u_{o}, v_{o}, w_{o}, f, z)$$

$$= \Phi^{*}(u_{o}, v_{o}, w_{o}, f, z + \Delta) - \frac{\Delta}{u_{o}} L^{*}(z + \Delta) + o(\Delta)$$

$$= f(u_{o}, v_{o}, w_{o}) - \frac{\Delta L^{*}}{u_{o}} (z + \Delta) + o(\Delta)$$
(8)

Thus, from Eqs. (5), (6), and (8), we may write

$$\int_{-\infty}^{0} \int_{-\infty}^{+\infty} \int_{-\infty}^{+\infty} R(u, v, w, u_{o}, v_{o}, w_{o}, z + \Delta) f(u_{o}, v_{o}, w_{o}) du_{o} dv_{o} dw_{o}$$

$$= \int_{-\infty}^{0} \int_{-\infty}^{+\infty} \int_{-\infty}^{\infty} R(u, v, w, u_{o}, v_{o}, w_{o}, z)$$

$$\left\{ f(u_{o}, v_{o}, w_{o}) - \frac{\Delta L}{u_{o}} (z + \Delta) \right\} \times du_{o} dv_{o} dw_{o} + \frac{1}{u} \Delta L^{+}(z) + o(\Delta)$$
(9)

We note that

$$\Delta L^{\pm}(z + \Delta) = \Delta L^{\pm}(z) + o(\Delta)$$

Equation (9) contains the gist of the invariant-imbedding argument. The Krook equation possesses the property that the perturbation distribution function and the linearized collision operator are reductible to a dependence on the single velocity variable, u, for linearized

If the linearized collision operator L is specified, then an integro-differential equation for R(u, v, w, u, v, w) is recovered.

steady and unsteady shear flows (boundary conditions permitting). This, of course, is a major virtue of this scattering model. Consequently, integrations over v_0 and w_0 are performed explicitly, and a significant reduction in the number of independent variables is obtained. An equation is obtained for the one-dimensional reflection function $R(u, u_0, \tau)$ by combining Eqs. (9) and (10).

$$L^{+} \equiv \left(\frac{\delta \delta^{+}}{dt}\right)_{\text{collisions}} = -\left\{\frac{\delta^{+}}{2} - \frac{1}{\sqrt{\pi}} \int_{0}^{\infty} e^{-u^{2}} \delta^{+} du - \frac{1}{\sqrt{\pi}} \int_{-\infty}^{\infty} e^{-u^{2}} \delta^{-} du_{o}\right\}$$

$$L^{-} \equiv \left(\frac{\delta \delta^{-}}{\delta t}\right)_{\text{collisions}} = -\left\{\delta^{-} - \frac{1}{\sqrt{\pi}} \int_{0}^{\infty} e^{-u^{2}} \delta^{+} du - \frac{1}{\sqrt{\pi}} \int_{-\infty}^{\infty} e^{-u^{2}} \delta^{-} du_{o}\right\}$$

$$(10)$$

Introducing Eq. (10) into Eq. (9), we obtain

$$\int_{-\infty}^{0} R(u, \bar{u}_{o}, z + \Delta) f(\bar{u}_{o}) du_{o} = \int_{-\infty}^{0} R(u, \bar{u}_{o}, z) \left\{ f(\bar{u}_{o}) + \frac{\Delta}{\bar{u}_{o}} \right\} \\
\times \left[\dot{r}(\bar{u}_{o}) - \frac{1}{\sqrt{\pi}} \int_{0}^{\infty} e^{-u^{1/2}} \int_{-\infty}^{0} R(u', u'_{o}, z) f(u'_{o}) du'_{o} du' \right] \\
- \frac{1}{\sqrt{\pi}} \int_{-\infty}^{0} f(u'_{o}) e^{-u'_{o}^{1/2}} du'_{o} \right\} d\bar{u}_{o} - \frac{1}{\sqrt{\pi}} \frac{1}{u} \Delta \int_{-\infty}^{0} R(u, \bar{u}_{o}, z) f(\bar{u}_{o}) d\bar{u}_{o} \\
+ \frac{1}{\sqrt{\pi}} \frac{1}{u} \Delta \int_{-\infty}^{\infty} e^{-u'_{o}^{1/2}} \int_{-\infty}^{0} R(u', u'_{o}, z) f(u'_{o}) du'_{o} du' \\
+ \frac{1}{\sqrt{\pi}} \frac{1}{u} \Delta \int_{-\infty}^{0} f(u'_{o}) e^{-u'_{o}^{1/2}} du'_{o} du'_{o} du'_{o}$$
(11)

where \bar{u}_0 , u_0^1 , and u^1 are dummy variables.

This relationship must hold for all source functions f(u). We select the particular value corresponding to a Dirac-delta-function source, i.e.,

$$f(u) = o(u - u_0)$$

Therefore,

$$R(u, u_{o}, \tau + \Delta) = R(u, u_{o}, z) + \frac{\Delta}{u_{o}} R(u, u_{o}, z)$$

$$- \Delta \int_{-\infty}^{0} R(u, u_{o}', z) \cdot \frac{1}{u_{o}' \sqrt{\pi}} \int_{0}^{\infty} e^{-u'^{2}} R(u', u_{o}) du' du'_{o}$$

$$- \Delta \int_{-\infty}^{0} R(u, u_{o}', z) : \sqrt{\pi} \frac{1}{u_{o}'} e^{-u^{2}} du'_{o} - \frac{\Delta}{u} R(u, u_{o}, z)$$

$$+ \frac{\Delta}{u \sqrt{\pi}} \int_{0}^{\infty} e^{-u'^{2}} R(u', u_{o}, z) du' + \frac{\Delta}{u \sqrt{\pi}} e^{-u^{2}} du'_{o} - \frac{\Delta}{u} R(u, u_{o}, z)$$
(12)

Letting $\Delta \rightarrow 0$, and introducing

$$R(u, u_0, z) = \frac{1}{\sqrt{\pi}} \frac{1}{\sqrt{u}} e^{-u_0^2} S(u, |u_0|, z)$$

in order to better obtain a symmetric form yields

$$\frac{dS}{dz}$$
 (u, | u_o|, z) + $\frac{S}{|u_o|}$ (u, |u_o|, z) + $\frac{S}{u}$ (u, |u_o|, z)

= 1 +
$$\frac{1}{\pi} \int_{0}^{\infty} \frac{e^{-u'^2}}{u'} S(u', |u_0|, z) du'$$

$$+\frac{1}{\pi}\int_{0}^{\infty}\frac{-u_{0}^{'2}}{\left|\frac{u_{0}^{'}}{u_{0}^{'}}\right|}S(u, \left|u_{0}^{'}\right|, z) du_{0}^{'}$$

$$+\frac{1}{\pi}\int_{0}^{\infty}\int_{0}^{\infty}\frac{e^{-u'^{2}}}{\left|\frac{u'}{o}\right|\cdot u'}S(u, |u'_{o}|, z)S(u', |u_{o}|, z) du' du'_{o}$$
 (13)

This equation was derived in Ref. 1 by particle-counting techniques.

METHOD II

A direct and powerful approach to the development of the invariantimbedding equations from the Boltzmann formulation uses simple perturbation techniques as the principal tool. We apply it here to a discrete and of the Boltzmann equation and then indicate the results for continuous models.

Discrete Model

We introduce a discrete model of the Boltzmann equation:

$$\frac{df_{1}^{+}}{dx} = K_{1}^{+}(f_{1}^{-}, f_{2}^{-} \dots f_{N}^{-}, f_{1}^{+}, f_{2}^{+} \dots f_{N}^{+})$$

$$\frac{df_{1}^{-}}{dx} = -K_{1}^{-}(f_{1}^{+}, f_{2}^{+}, \dots f_{N}^{+}, f_{1}^{-}, f_{2}^{-} \dots f_{N}^{-})$$
(14)

The K_1^{\pm} 's are related to discrete versions of the collision operator and are not restricted solely to linear operators on the f_1^{\pm} 's. Discrete models of transport equations have been considered for many years in neutron and radiative transfer, but it is only in recent years that attempts have been made to utilize them in kinetic theory. We note that the subscript i refers to a particular value of the velocity vector, \vec{C}_i , defined by $\vec{C}_i = (u_{i_U}, v_{i_V}, w_{i_W})$, the f_i^{\pm} refers to the distribution function of molecules having positive u components, and the f_i^{\pm} 's refer to molecules having negative u components where u is positive in the direction of increasing x.

The discrete model arises from the consideration of a discrete set of molecular velocities, \vec{C}_i , rather than a continuous set. Finite volumes in velocity space $\Delta \vec{C}_i = \left(\Delta u_{i_u}\right) \left(\Delta v_{i_v}\right) \left(\Delta w_{i_w}\right)$ are utilized, rather than infinitesmal elements $d\vec{c} = du \ dv \ dw$; integrals are replaced by sums, and a distribution function corresponding to each of the states \vec{C}_i is sought.

We consider boundary conditions resulting from an arbitrary source of molecules at x = z and a sink for molecules at x = 0. A somewhat improbable physical interpretation of these boundary conditions is that the wall at x = z evaporates molecules into the flow and possesses a unit sticking coefficient, while the wall at x = 0 condenses all molecules which impinge on it. For linear problems, as discussed earlier, this standard set of boundary conditions is sufficient to account for other more interesting surface-particle interactions.

These boundary conditions are

$$\mathbf{x} = 0 \qquad \mathbf{f}_{\mathbf{i}}^{+} = 0$$

$$\mathbf{x} = \mathbf{z} \qquad \mathbf{f}_{\mathbf{i}}^{-} = \gamma_{\mathbf{i}}^{-} \qquad (15)$$

Let the solution of Eq. (14) on the interval 0, z, subject to the source γ_i at x = z, be indicated by f_i^+ , f_i^- .

Similarly, let the solution of Eq. (15) on the interval 0, $z + \Delta$, subject to the source γ_1 at $x = z + \Delta$, be indicated by F_1^+ , F_2^- .

Thus we may write

$$\frac{df_{1}^{+}}{dx} = k_{1}^{+}(f_{1}^{+}, f_{2}^{+} \dots f_{N}^{+}, f_{1}^{-}, f_{2}^{-} \dots f_{N}^{-})$$

$$\frac{df_{1}^{-}}{dx} = -k_{1}^{-}(f_{1}^{+}, f_{2}^{+} \dots f_{N}^{+}, f_{1}^{-}, f_{2}^{-} \dots f_{N}^{-})$$
(16)

where i runs from 1 to N with boundary conditions

$$f_{i}^{\dagger}(0) = 0$$

$$f_{i}^{\dagger}(z) = \gamma_{i}^{\dagger}$$
(17)

and

$$\frac{dF_{1}^{+}}{dx} = K_{1}^{+}(F_{1}, F_{2} \dots F_{N}, F_{1}^{-}, F_{2}^{-} \dots F_{N}^{-})$$

$$\frac{dF_{1}^{-}}{dx} = -K_{1}^{-}(F_{1}^{+}, F_{2}^{+} \dots F_{N}^{+}, F_{1}^{-}, F_{2}^{-} \dots F_{N}^{-})$$
(18)

where the k_{1}^{\pm} is the same collision operator as the k_{1}^{\pm} , where

$$F_{i}^{+}(0) = 0$$

$$F_{i}^{-}(z + \Delta) = \gamma_{i}^{-}$$
(19)

We seek perturbation solutions to Eq. (18) in the form

$$\mathbf{F}_{1}^{+} = \mathbf{f}_{1}^{-} + \Delta \cdot \mathbf{r}_{1}^{+} + o(\Delta)$$

$$\mathbf{F}_{1}^{-} = \mathbf{f}_{1}^{-} + \Delta \cdot \mathbf{w}_{1}^{-} + o(\Delta)$$
(20)

- "

The boundary conditions on r_i^+ , w_i^- can be obtained as follows:

$$r_{i}^{+}(0) = 0$$
 at $x = 0$ (21a)

At x = z, we expand in a Taylor series to obtain

$$F_{\underline{i}}(z) = F_{\underline{i}}(z + \Delta) - \Delta \cdot \frac{\partial F_{\underline{i}}}{\partial x} (z + \Delta) + o(\Delta)$$
 (21b)

or

$$f_{i}(z) + \Delta \cdot w_{i}(z) = \gamma_{i} - \Delta \cdot \frac{\partial f_{i}}{\partial x}(z) + o(\Delta)$$

or

$$\mathbf{w}_{1}^{-}(\mathbf{z}) = k_{1}^{-}(\mathbf{f}_{1}^{+}(\mathbf{z}), \mathbf{f}_{2}^{+}(\mathbf{z}) \dots \mathbf{f}_{N}^{+}(\mathbf{z}), \mathbf{f}_{1}^{-}(\mathbf{z}), \mathbf{f}_{2}^{-}(\mathbf{z}) \dots \mathbf{f}_{N}^{-}(\mathbf{z}))$$
 (21c)

The linear differential equations for the perturbations r_i^+ , w_i^- are obtained by substitution of Eq. (20) into Eq. (18).

$$\frac{\partial \mathbf{r}_{i}^{+}}{\partial \mathbf{x}} = \sum_{j=1}^{N} \frac{\partial \mathbf{k}_{i}^{+}}{\partial \mathbf{f}_{j}^{+}} \mathbf{r}_{j}^{+} + \sum_{j=1}^{N} \frac{\partial \mathbf{k}_{i}^{+}}{\partial \mathbf{f}_{j}^{-}} \mathbf{w}_{j}^{-}$$

$$\frac{\partial \mathbf{w}_{i}^{-}}{\partial \mathbf{x}} = -\sum_{j=1}^{N} \frac{\partial \mathbf{k}_{i}^{-}}{\partial \mathbf{f}_{j}} \mathbf{r}_{j}^{+} + \sum_{j=1}^{N} \frac{\partial \mathbf{k}_{i}^{-}}{\partial \mathbf{f}_{j}^{-}} \mathbf{w}_{j}^{-}$$
(22)

The solution to this set of equations, subject to boundary conditions in Eqs. (21), is

$$r_{i}^{+}(x) = \sum_{j=1}^{N} \frac{\partial f_{i}}{\partial \gamma_{j}^{-}} \cdot k_{j}^{-}(f_{1}^{+}(z), f_{2}^{+}(z) \dots f_{N}^{+}(z), f_{1}^{-}(z), f_{2}^{-}(z) \dots f_{N}^{-}(z))$$

$$w_{i}^{-}(x) = \sum_{j=1}^{N} \frac{\partial f_{i}^{-}}{\partial \gamma_{j}^{-}} k_{j}^{-}(f_{1}^{+}(z), f_{2}^{+}(z) \dots f_{N}^{+}(z), f_{1}^{-}(z), f_{2}^{-}(z) \dots f_{N}^{-}(z))$$
(23)

which can be verified by substitution.

At $x = z + \Delta$, a Taylor series expansion results in

$$F_{i}^{+}(z + \Delta) = F_{i}^{+}(z) + \frac{\Delta \partial F_{i}^{+}(z)}{\partial x} + o(\Delta)$$

$$= f_{i}^{+}(z) + \Delta r_{i}^{+}(z)$$

$$+ \Delta k_{i}^{+}(f_{1}^{+}(z), f_{2}^{+}(z) \dots f_{N}^{+}(z), f_{1}^{-}(z), f_{2}^{-}(z) \dots f_{N}^{-}(z)) + o(\Delta)$$

$$(24)$$

At this point in the analysis it is appropriate to introduce a reflection function that must differ essentially from the reflection function of the linear problem considered earlier. In the linear problem, it was assumed that the distribution function for particles impliging on the boundary at x = z was a linear functional of the distribution function for particles emitted from that boundary. definition of the reflection function in Eqs. (3) and all the succeeding analysis was engendered by the linearity of the collision operator. Thus it was possible to consider a delta-function incident flux of molecules and construct from this the solution for any incident flux. In this nonlinear version of the same problem it is no longer possible to uncouple the el. of a particular source distribution at the boundary at x = z from the resulting flux of particles incident upon that boundary. We introduce the function $R_i^*(z, \gamma_1, \gamma_2 \dots \gamma_N)$ which is the distribution function of particles incident upon the boundary

at x=z due to the particular distribution of incident particles at x=z emitted by that boundary, γ_1 , γ_2 ... γ_N .

$$R_{i}^{*}(z, \gamma_{1}^{-}, \gamma_{2}^{-} \dots \gamma_{N}^{-}) \equiv f_{i}^{+}(z,)$$

$$R_{i}^{*}(z + \Delta, \gamma_{1}^{-}, \gamma_{2}^{-} \dots \gamma_{N}^{-}) \equiv f_{i}^{+}(z + \Delta)$$

and

$$\frac{\partial R_{i}^{*}}{\partial z} = \lim_{\Delta \to 0} \frac{F_{i}(z + \Delta) - f_{i}(z)}{\Delta}$$

The resulting equation for $R_{i}^{*}(z)$ is then

$$\frac{\partial R_{i}^{*}}{\partial \mathbf{z}} (\mathbf{z}, \, \mathbf{y}_{1}^{-}, \, \mathbf{y}_{2}^{-} \dots \, \mathbf{y}_{N}^{-}) = \sum_{j=1}^{N} \frac{\partial R_{i}(\mathbf{x}, \, \mathbf{y}_{1}^{-}, \, \mathbf{y}_{2}^{-} \dots \, \mathbf{y}_{N}^{-})}{\partial \mathbf{y}_{j}^{-}} \\
\times k_{j}^{-} (R_{1}^{*}(\mathbf{z}), \, R_{2}^{*}(\mathbf{z}) \dots \, R_{N}^{*}(\mathbf{z}), \, \mathbf{y}_{1}^{-}, \, \mathbf{y}_{2}^{-} \dots \, \mathbf{y}_{N}^{-}) \\
+ k_{i}^{+} (R_{1}(\mathbf{z}), \, R_{2}(\mathbf{z}) \dots \, R_{N}(\mathbf{z}), \, \mathbf{y}_{1}^{-}, \, \mathbf{y}_{2}^{-} \dots \, \mathbf{y}_{N}^{-}) \tag{25}$$

Continuous Model

If a continuous version of Eqs. (16) and (17) is considered,

$$\frac{df^{+}}{dx}(\vec{C}^{+}, x) = K^{+}(\vec{C}^{+})$$

$$\frac{df^{-}}{dx}(\vec{C}^{-}, x) = -K^{-}(\vec{C}^{+})$$

$$f^{+}(\vec{C}^{+}, 0) = 0$$

$$f^{-}(\vec{C}^{-}, z) = \gamma(\vec{C}^{-})$$
(26)

where the $K(\overrightarrow{C}^{\pm})$ are functionals of the distribution functions f^{\pm} , and \overrightarrow{C} takes on all values of the velocity vector possessing a component in the direction of decreasing x, while \overrightarrow{C}^{\pm} takes on all values possessing a component in the direction of increasing x. The results of the previous section may be extended formally to give

$$\frac{\partial R^{*}}{\partial \mathbf{z}} (\gamma(\vec{C}), \vec{C}, \mathbf{z}) = \int_{\vec{C}}^{*} \frac{\delta R^{*}}{\delta \gamma} (\gamma(\vec{\xi}), \vec{C}, \mathbf{z}) K^{-}(\vec{\xi}^{-}, \mathbf{z}) d\vec{\xi}^{-} + K^{+}(\vec{C}^{+}, \mathbf{z})$$
(27)

where $\frac{\partial R^*}{\delta \gamma}$ is the functional partial derivative, defined by

$$\lim_{\Delta \to 0} \frac{R^*(Y(\vec{C}) + \Delta K^*(\vec{C}), \vec{C}, z) - R^*(Y(\vec{C}), \vec{C}, z)}{\Delta}$$

$$= \int_{-\infty}^{0} \int_{-\infty}^{+\infty} \int_{-\infty}^{+\infty} R^{*}(Y(\vec{C}_{0}), \vec{C}, z) \cdot K^{-}(\vec{C}_{0}, z) d\vec{C}_{0}$$
 (28)

If K is a linearized collision operator, we introduce a reflection function $R^*(v, v, w, u_o, v_o, z)$ such that

$$R^{*}(\gamma(\vec{C}_{O}), \vec{C}, z) = \int_{-\infty}^{0} \int_{-\infty}^{-\infty} \int_{-\infty}^{+\infty} R^{*}(u, v, w, u_{O}, v_{O}, w_{O}, z)$$

$$\times \gamma(u_{O}, v_{O}, w_{O}) du_{O} dv_{O} dw_{O}$$
(29)

and note that

$$\frac{\delta R^*}{\delta y} = R^*(u, v, w, u_o, v_c, w_o, z)$$
 (30)

Substituting Eqs. (29) and (30) in Eq. (27), we recover the equivalent of Eq. (9):

$$\frac{\partial}{\partial z} = \int_{-\infty}^{0} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} R^{*}(u, v, w, u_{o}, v_{o}, w_{o}, z) Y(u_{o}, v_{o}, w_{o}) du_{o} dv_{o} dw_{o}$$

$$= - \int_{-\infty}^{0} \int_{-\infty}^{+\infty} \int_{-\infty}^{+\infty} R^{*}(u, v, w, u'_{o}, v'_{o}, w'_{o}, z) \frac{L^{-}}{u'_{o}}$$

$$\times (u'_{o}, v'_{o}, w'_{o}, z) du'_{o} dv'_{o} dw'_{o} + \frac{1}{u} L^{+}(u, v, w, z)$$
(31)

The initial condition is always $R^{\dagger}(\vec{C}, \vec{C}_{0}, 0) \equiv 0$, stemming from the observation that there are no collisions to reflect particles from \vec{C}_{0} to \vec{C} if $\tau = 0$.

REFERENCES

- 1. Aroesty, J., et al., "Invariant Imbedding and Rarefied Gas Dynemics," Proc. Nat. Acad. Sci., Vol. 50, August 1963, pp. 222-226.
- 2. Aroesty, J., et al., An Invariant Imbedding Approach to Rarefied Couette Flows, AIAA Preprint 64-34, January 1964.
- 3. Bailey, P. B., "A Rigorous Derivation of Some Invariant Imbedding Equations of Transport Theory," <u>J. Math. Anal. & Appl.</u>, Vol. 8, February 1964, pp. 144-169.
- 4. Wing, G. M., "Invariant Imbedding and Transport Theory A Unified Approach," J. Math. Anal. & Appl., Vol. 2, 1961, pp. 277-292.
- 5. Bellman, R., and R. Kalaba, "On the Fundamental Equations of Invariant Imbedding," <u>Proc. Nat. Acad. Sci.</u>, Vol. 47, 1961, pp. 336-338.

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tions. It is shown how both	linear and	Physics			
nonlinear collision operator	rs may be con-	ĺ			
sidered, for both discrete a	and continuous				
versions of the original Bol	ltzmann formu-				
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